

Rayleigh-Taylor Unstable Flames

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Type Ia Supernovae

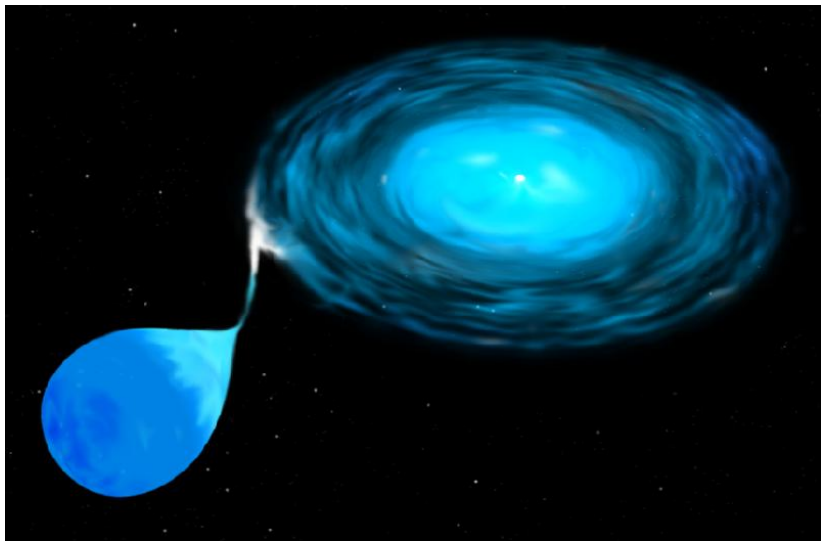


Image: NASA

Deflagration to Detonation Transition (DDT)

In order to get the correct explosion properties the SN Ia flame must become supersonic.

- ▶ Deflagration: A subsonic flame
- ▶ Detonation: A supersonic flame with an associated shock wave

Two ways for this to occur:

- ▶ Zeldovich Gradient Mechanism
- ▶ Flame exceeds the sound speed because fluid instabilities (Rayleigh-Taylor instability) increase the surface area of the flame

Rayleigh-Taylor Instability

The source of wrinkling: the Rayleigh-Taylor instability

- ▶ The fuel is more dense than the ash
- ▶ The flame propagates upward against the direction of gravity

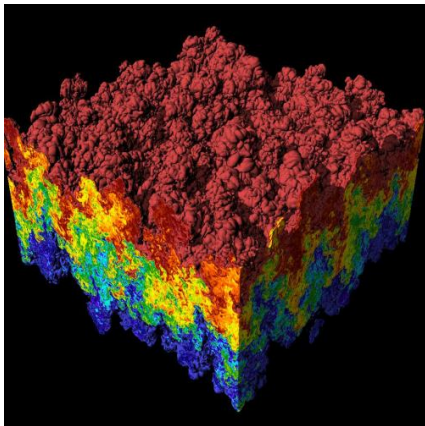


Image: LLNL

Some important questions:

How do flows generated by the flame interact with the flame front?

- ▶ What causes changes in the surface area of the flame?
- ▶ At what G does the Rayleigh-Taylor instability become important?
- ▶ How much can Rayleigh-Taylor driven turbulence wrinkle the flame front?

Difficulties for Simulations

1. **Current Supercomputers are unable to resolve the flame and the whole white dwarf at the same time.**
 - ▶ Size of White Dwarf = $7 * 10^8$ cm, Width of Flame = .01 – 1 cm
 - ▶ Subgrid Models for flame behavior are necessary

What kind of subgrid model is actually appropriate?

- ▶ Flame Speed = Rayleigh-Taylor speed
- ▶ Flame Speed determined by Kolmogorov Turbulence

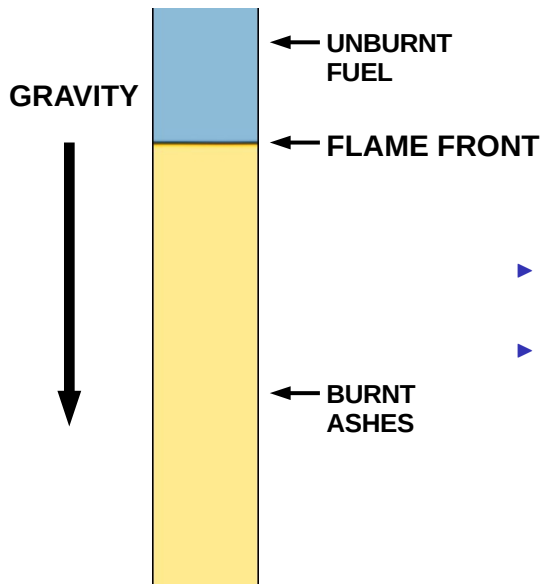
2. **Full Reaction Networks are very stiff and difficult to integrate**

Study Flames Directly

Avoid these problems and study a very simplified case:

1. A flame in a rectangular computational domain
2. Constant gravity
3. Use a simple, model reaction

Flame in a Gravitational Field



- ▶ The flame is Rayleigh-Taylor unstable.
- ▶ Code: Nek5000 (P. Fischer, ANL)

Important Parameters

Non-Dimensional Gravity:

$$G = g \left(\frac{\Delta\rho}{\rho_o} \right) \frac{\delta}{s_o^2}$$

Prandtl Number:

$$Pr = \frac{\nu}{\kappa} = 1$$

Non-Dimensional Box Size:

$$L = \frac{l}{\delta} = 128$$

What happens when gravity is increased?

Transition-to-turbulence type problem

- ▶ Look for low-dimensional dynamical systems and simple bifurcations when G is small
- ▶ Consider the effects of turbulence when G is large

Dynamical System

A **dynamical system** consists of a state space plus a rule for time evolution in that space.

$$x = \Theta$$
$$y = \dot{\Theta}$$

$$\dot{y} = -\frac{g}{L}\sin(x)$$
$$\dot{x} = y$$

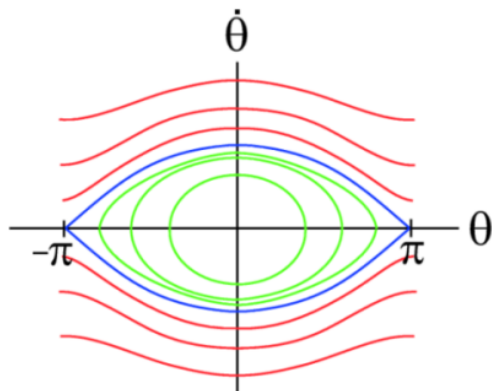


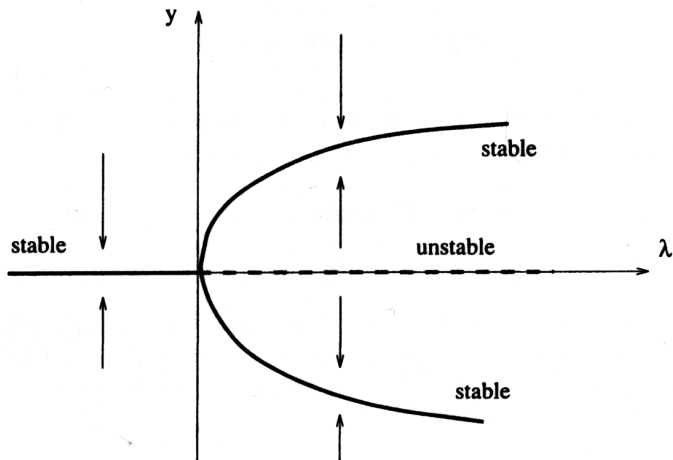
Image: S. Shadden

Bifurcations

A **bifurcation** is a change in a system as a **control parameter** is increased. Example:

$$\dot{y} = \lambda y - y^3$$

λ is the **control parameter**



Low G Systems

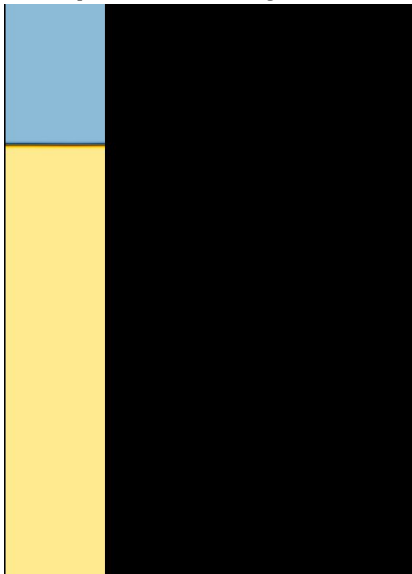
General Strategy:

- ▶ Pick an observable to model (examples: velocity at a point, flame speed)
- ▶ Treat the observable as part of an underlying low-dimensional dynamical system.

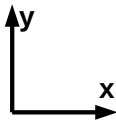
Spatial behaviors can be understood temporally

Flame: Fixed Point Example ($G = 0.001$)

Temp. **Vx** **Vy** **Vorticity**

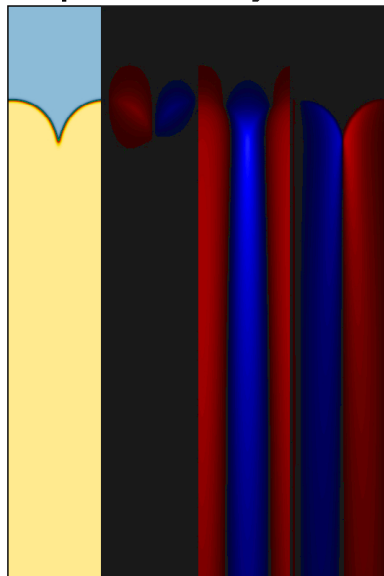


- ▶ The flame moves upward at a constant speed.

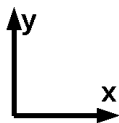


Cusped Flame Front, Stable Rolls ($G = 0.17$)

Temp. V_x V_y Vorticity

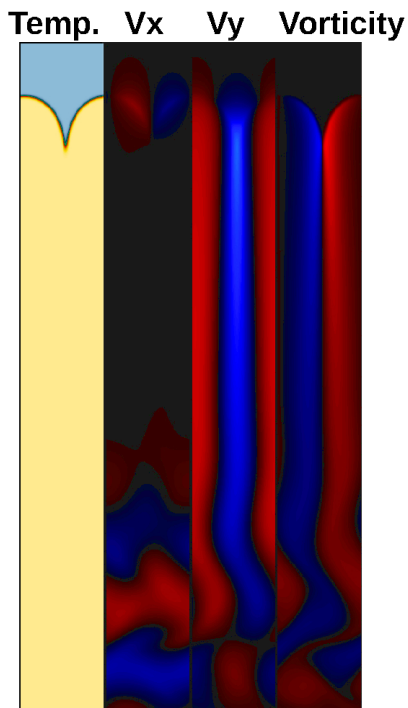


- ▶ Flame front becomes **cusped**.
- ▶ **Stable rolls** attach to the flame front
- ▶ Flame speed is still constant



Unstable Rolls ($G = 0.24$)

- ▶ **Shear instability**
destabilizes the rolls
- ▶ **Vortex Shedding** begins!
- ▶ Flame speed is still constant



Is there a low-dimensional model that describes the vortex shedding?

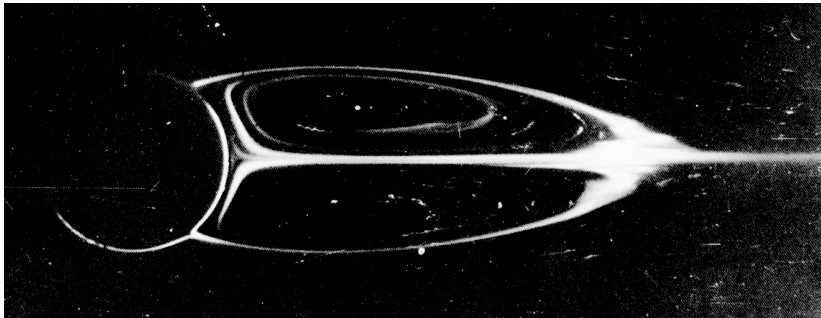


Image: Van Dyke

- ▶ Vortex shedding creates periodic behavior: **Hopf bifurcation**
- ▶ Similar system: flow past a cylinder (von Karman vortex street)
- ▶ Cylinder: Shedding at near critical Re governed by the Landau equation (Strykowski and Sreenivasan 1986,1987,1990)

Landau Equation: Governs shedding behind the flame

The Landau equation is satisfied in the vortex forming region behind the flame.

$$\frac{dA}{dt} = aA - \frac{1}{2}c|A|^2A$$

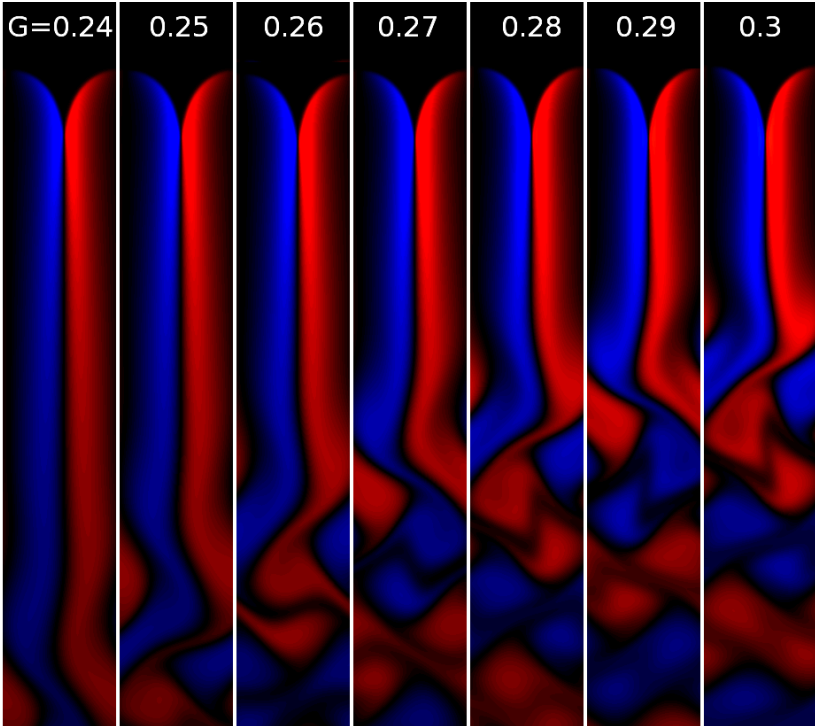
- ▶ In this case, choose V_x as A
- ▶ The beginning of periodic motion is a Hopf bifurcation.
- ▶ $G_{cr} \approx .22$

Vortex shedding is can be modelled by a time-dependent process with a secondary spatial dependence.

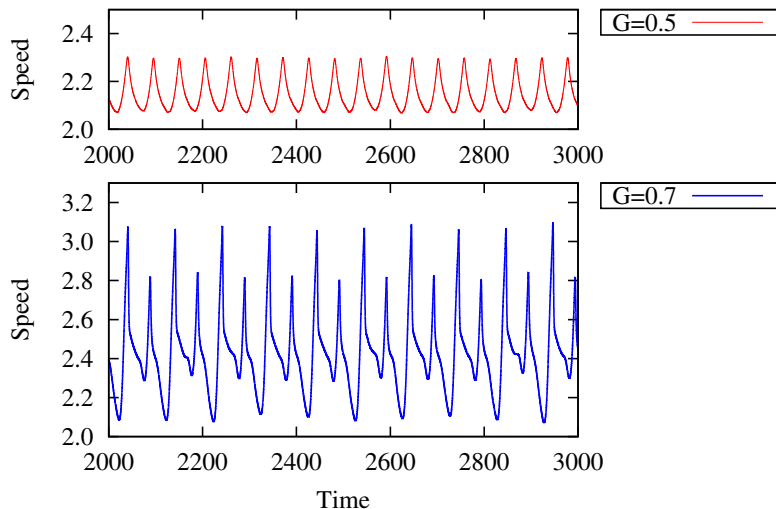
Time Series Analysis of the Flame Speed

Observe the flame speed to learn about the underlying “dynamical system”

- ▶ When vortex shedding moves close enough to the flame front, the flame speed begins to oscillate. ($G \approx 0.3$)
- ▶ Period Doubling: Left/Right Symmetry Breaks ($G \approx 0.6$)
- ▶ Torus Bifurcation: Cusp Breaks ($G \approx 0.85$)

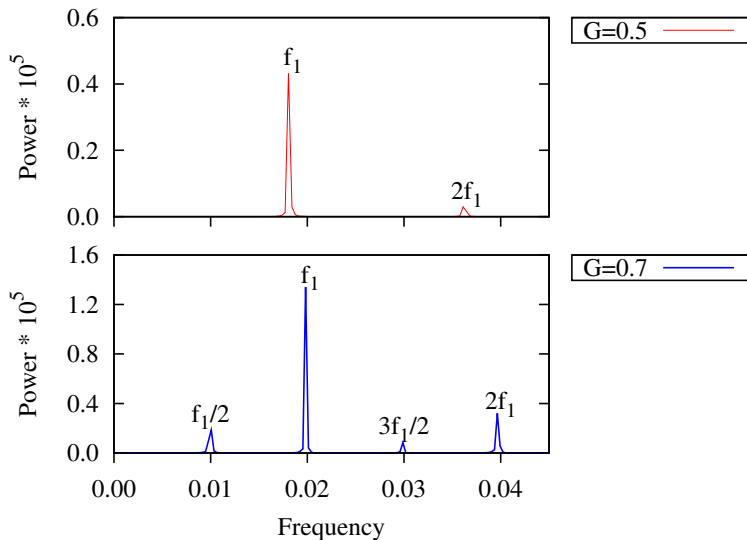


Period Doubling

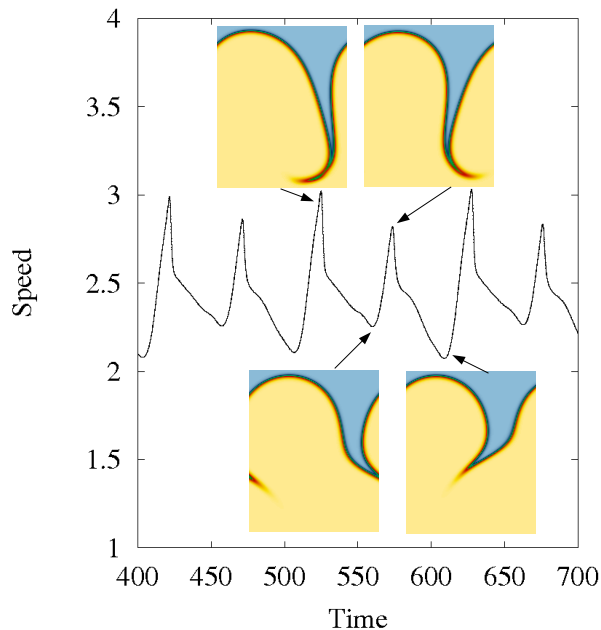


- ▶ Between $G = 0.5$ and $G = 0.7$ a period doubling occurs.

Period Doubling: Power Spectra

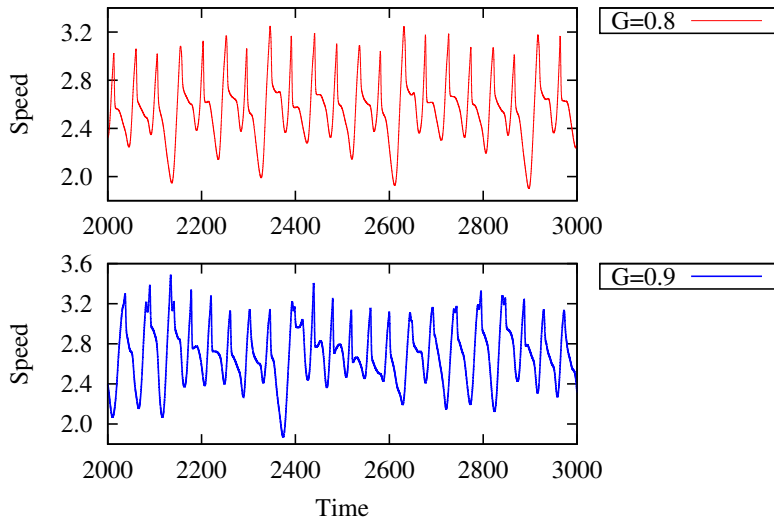


Period Doubling- Left/Right Symmetry Breaks



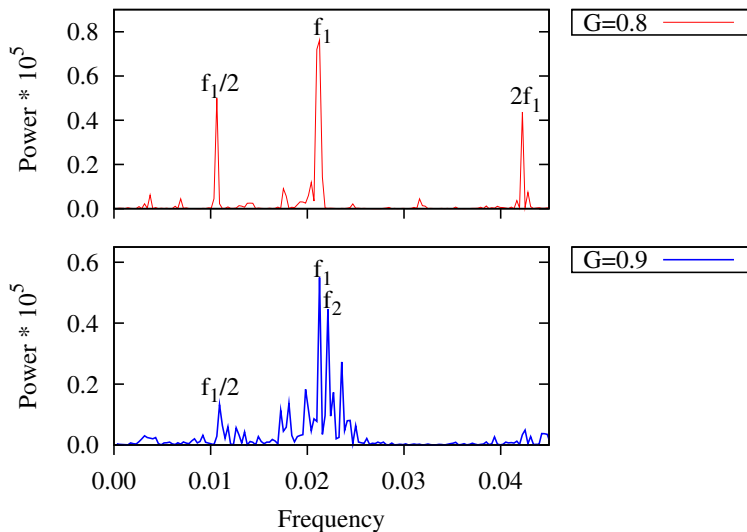
- ▶ **Base Period:**
Up/down
motion of the
flame
- ▶ **Doubled
Period:**
Side-to-side
motion

Torus bifurcation



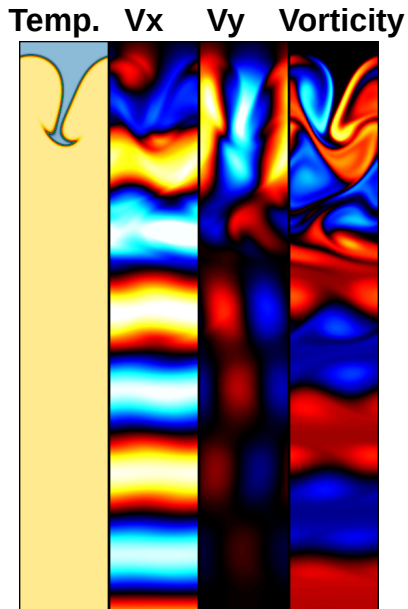
- ▶ An extra, incommensurate frequency develops.

Torus bifurcation: Power Spectra

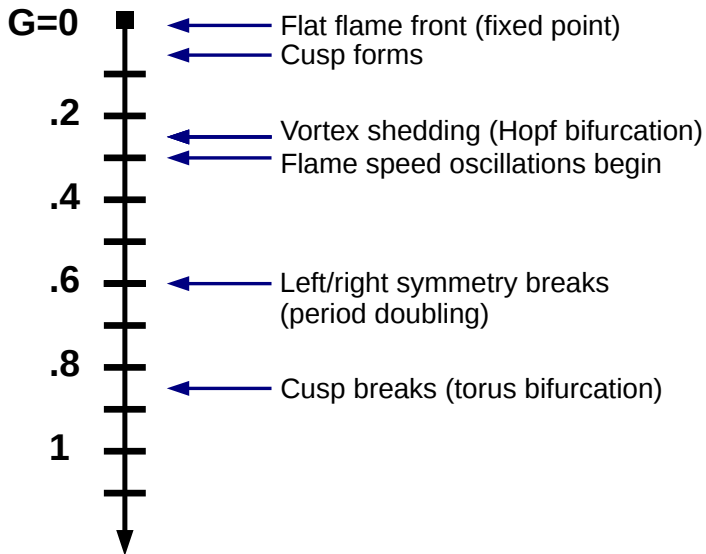


Torus Bifurcation: Cusp Breaking ($G = 0.9$)

- ▶ New Frequency introduced by **cusp breaking**
- ▶ The cusp is broken when the **Rayleigh-Taylor instability** begins to overwhelm burning.



Summary for $G \leq 1$



Conclusions for low- G flames

Dynamical systems theory gives simple, but powerful, models for understanding complex-looking flame behaviors.

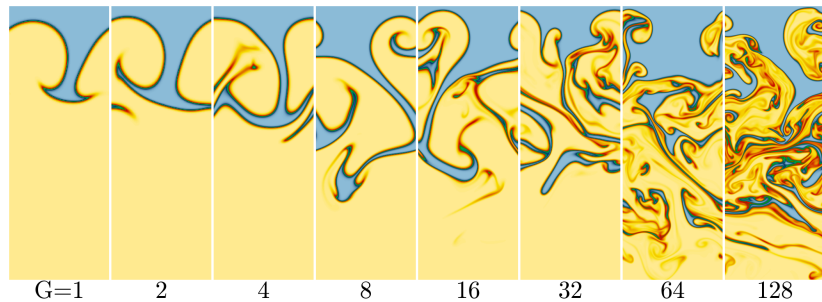
What causes changes in the surface area of the flame?

- ▶ At low G , the cusp creates vortex rolls which become unstable by a shear instability. These rolls affect the flame surface from behind.
- ▶ The flame front is disturbed by material behind it even at low values of G .

At what G does the Rayleigh-Taylor instability become important?

- ▶ R-T becomes directly important at $G = 0.9$

High G Flames ($G \geq 1$)

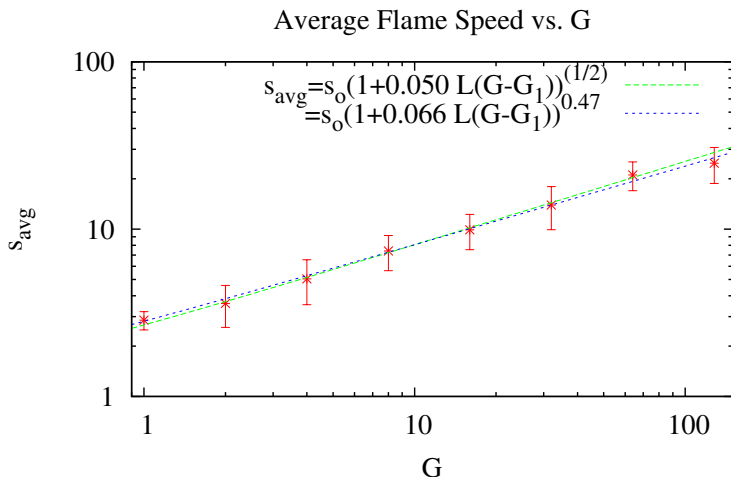


Two important questions:

1. What is the flame speed?
2. How well is turbulence able to wrinkle the flame front?

Flame Speed Scaling

The average flame speed scales as $s \propto \sqrt{GL}$.



Subgrid models should be based on the Rayleigh-Taylor flame speed!

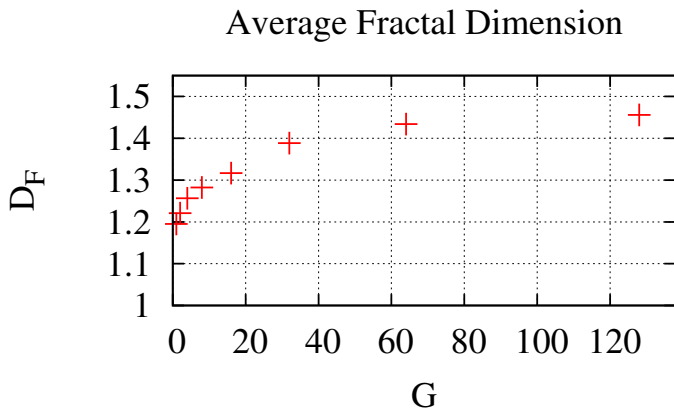
Box-Counting (Fractal) Dimension

$$D_{frac} = - \lim_{\epsilon \rightarrow 0} \frac{\log(N(\epsilon))}{\log(\epsilon)}$$



(Mandelbrot 1967)

Measuring Flame Wrinkling- Fractal Dimension



- ▶ The flame front is wrinkled by turbulence, but the amount of wrinkling levels off.
- ▶ **Claim: At high values of G , large-scale stretching of the flame by the RT instability controls the flame dynamics.**

Fractal Dimension Model for the Flame Speed

The flame speed is proportional to the flame area:

$$\frac{s}{s_0} = \frac{L_f}{L}$$

The length of the flame is determined by:

- ▶ Large-scale Rayleigh-Taylor stretching
- ▶ Wrinkling at all scales by turbulence

What are the contributions from each of these effects?

A model for the flame area:

$$L_f = L_{RT}(G) * \left(\frac{L}{\eta(G)} \right)^{D_F - 1}$$

L = the size of the largest eddies (the box size)

η = the size of the smallest eddies (the Kolmogorov cutoff scale)

Some useful relations:

- ▶ For large values of G , $D_F \rightarrow 1.5$.
- ▶ For 2D turbulence, $\eta = L * Re^{-1/2}$.
- ▶ $Re = L(1 + 0.2L(G - G_1))^{0.56}$

Altogether this gives:

$$L_f \propto L_{RT}(G) * G^{0.14}$$

We already know that $s \propto \sqrt{GL}$ so:

- ▶ Rayleigh-Taylor stretching scales as: $G^{0.36}$
- ▶ Turbulent wrinkling scales as: $G^{0.14}$

Large-scale Rayleigh-Taylor stretching dominates the flame speed for high values of G .

Conclusions

- ▶ At low G , flames can be described by simple, low-dimensional bifurcation models.
- ▶ The initial disturbance of the flame surface is due to a shear instability behind the flame front.
- ▶ At large G , the Rayleigh-Taylor instability controls the burning rate by stretching the flame front.
- ▶ Subgrid models should be based on the Rayleigh-Taylor flame speed.
- ▶ The DDT transition is not possible for flames under these conditions.

DNS/LES computational fluid dynamics solver

Developers: Paul Fischer (chief architect), Aleks Obabko, James Lottes, Stefan Kerkemeier, Katie Heisey

Features:

- ▶ Spectral Element Code
- ▶ Incompressible or low-Mach number
- ▶ Very fast and efficient
- ▶ Low memory use
- ▶ Scales up to 100,000 processors
- ▶ Works very well with solid boundaries

Website: <http://nek5000.mcs.anl.gov>

Flame Simulations

Parameters-WD

| ρ_9 | G | L | GL |
|----------|------------------|-----------------|--------------|
| 10 | $3.1 * 10^{-13}$ | $7.9 * 10^{12}$ | 2.44 |
| 8 | $6.4 * 10^{-13}$ | $6.1 * 10^{12}$ | 3.9 |
| 6 | $1.9 * 10^{-12}$ | $4 * 10^{12}$ | 7.6 |
| 4 | $9.5 * 10^{-12}$ | $2 * 10^{12}$ | 19.5 |
| 2 | $2.1 * 10^{-10}$ | $5.4 * 10^{11}$ | 114 |
| 1 | $6.5 * 10^{-9}$ | $1.4 * 10^{11}$ | 910 |
| .5 | $2.1 * 10^{-7}$ | $3.6 * 10^{10}$ | 7500 |
| .2 | $6.8 * 10^{-5}$ | $4.9 * 10^9$ | $3.3 * 10^5$ |
| .1 | $2 * 10^{-2}$ | $1.2 * 10^8$ | $2.5 * 10^6$ |
| .05 | $8.7 * 10^{-1}$ | $4.3 * 10^7$ | $3.8 * 10^7$ |
| .01 | $4.5 * 10^2$ | $1.2 * 10^7$ | $5.2 * 10^9$ |

Parameters-Simulations

| G | L | GL |
|----------|----------|-----------|
| .25 | 128 | 32 |
| .5 | 128 | 64 |
| 1 | 128 | 128 |
| 2 | 128 | 256 |
| 4 | 128 | 512 |
| 8 | 128 | 1024 |
| 16 | 128 | 2048 |
| 32 | 128 | 4096 |
| 64 | 128 | 8192 |
| 128 | 128 | 16384 |

The Equations

- ▶ Use the Boussinesq Approximation

Equations

Navier-Stokes Equation:

$$\frac{D\mathbf{u}}{Dt} = - \left(\frac{1}{\rho_o} \right) \nabla p + \frac{\rho \mathbf{g}}{\rho_o} + \nu \nabla^2 \mathbf{u}$$

Heat Equation:

$$\frac{DT}{Dt} = \kappa \nabla^2 T + R(T)$$

Continuity Equation:

$$\nabla \cdot \mathbf{u} = 0$$

Bistable Reaction:

$$R(T) = 2T^2(1 - T)$$

Definitions

Laminar Flame Speed:

$$s_o = \sqrt{\alpha \kappa}$$

Flame Width:

$$\delta = \sqrt{\frac{\kappa}{\alpha}}$$

κ is the thermal diffusivity

$\frac{1}{\alpha}$ is the reaction time

Non-Dimensional Equations

Equations

Navier-Stokes Equation:

$$\frac{D\mathbf{u}}{Dt} = - \left(\frac{1}{\rho_o} \right) \nabla p + GT + Pr \nabla^2 \mathbf{u}$$

Heat Equation:

$$\frac{DT}{Dt} = \nabla^2 T + 2T^2(1 - T)$$

Continuity Equation:

$$\nabla \cdot \mathbf{u} = 0$$

Control Parameters

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